

AMENDMENTS TO THE CLAIMS

The following listing of claims will replace all prior versions and listings of claims in the application.

LISTING OF CLAIMS

1. (currently amended) A method of running a cable from an interior of an aircraft to an exterior of the aircraft through an opening in a fuselage of the aircraft, the method comprising:
 - (a) inserting [[a]] an integral single piece guide member through an opening in a fuselage of an aircraft;
 - (b) forming a first seal between said guide member and said opening;
 - (c) inserting running a cable through an opening in said guide member so that said cable extends through said fuselage of said aircraft;
 - (d) forming a second seal between said guide member and said cable;
 - (e) applying a vacuum to said first and second seals; and
 - (f) observing a level of said vacuum after a period of time.
2. (original) The method of claim 1, wherein prior to performing step (c), step (b) further comprises applying a vacuum to said first seal and observing a level of said vacuum after a period of time.

3. (original) The method of claim 2, wherein step (b) further comprises applying a vacuum to said first seal by placing a body of a testing tool having a cavity over said opening with said cavity communicating with said opening and with said first seal and forming a vacuum in said cavity with a vacuum source.

4. (currently amended) ~~The method of claim 2, further comprising forming said vacuum in said cavity with a vacuum source~~ A method of running a cable from an interior of an aircraft to an exterior of the aircraft through an opening in a fuselage of the aircraft, the method comprising:

- (a) inserting a guide member through an opening in a fuselage of an aircraft;
- (b) forming a first seal between said guide member and said opening, applying a vacuum to said first seal by placing a body of a testing tool having a cavity over said opening with said cavity communicating with said opening and with said first seal and forming a vacuum in said cavity with a vacuum source that communicates with said cavity through said opening in said guide member and observing a level of said vacuum after a period of time;
- (c) running a cable through an opening in said guide member so that said cable extends through said fuselage of said aircraft;
- (d) forming a second seal between said guide member and said cable;
- (e) applying a vacuum to said first and second seals; and
- (f) observing a level of said vacuum after a period of time.

5. (original) The method of claim 1, wherein step (e) further comprises positioning a body of a testing tool having a cavity over said opening with said cavity communicating with said opening and with said first and second seals and forming a vacuum in said cavity with a vacuum source.

6. (currently amended) A method of running a cable from an interior of an aircraft to an exterior of the aircraft through an opening in a fuselage of the aircraft, the method comprising:

- (a) inserting running a cable through an opening in a fuselage of an aircraft with said cable extending from an interior of said aircraft to an exterior of said aircraft;
- (b) forming a seal between said cable and said opening by applying a sealant directly to said cable and said opening;
- (c) applying a vacuum to said seal; and
- (d) observing a level of said vacuum after a period of time.

7. (original) The method of claim 6, wherein (c) further comprises positioning a body of a testing tool having a cavity over said opening with said cavity communicating with said opening and with said seal and forming a vacuum in said cavity with a vacuum source.

8. (original) The method of claim 6, wherein said cable is an antenna cable.

9. (original) The method of claim 6, wherein said cable has a diameter greater than about 0.5 inches.

10. (currently amended) A method of forming a conduit for running a cable from an interior of an aircraft to an exterior of the aircraft through an opening in a fuselage of the aircraft, the method comprising:

- (a) inserting [[a]] an integral single piece guide member having an opening therein through an opening in a fuselage of an aircraft thereby forming a conduit for a cable;
- (b) forming a first seal between said guide member and said opening;
- (c) applying a vacuum to said first seal; and
- (d) observing a level of said vacuum after a period of time.

11. (original) The method of claim 10, wherein (c) includes positioning a body of a testing tool having a cavity over said opening with said cavity communicating with said opening and with said first seal and forming a vacuum in said cavity with a vacuum source.

12. (currently amended) ~~The method of claim 11, further comprising A~~
method of forming a conduit for a cable from an interior of an aircraft to an exterior of
the aircraft through an opening in a fuselage of the aircraft, the method comprising:

- (a) inserting a guide member through an opening in a fuselage of an
aircraft thereby forming a conduit for a cable;
- (b) forming a first seal between said guide member and said opening;
- (c) applying a vacuum to said first seal by positioning a body of a
testing tool having a cavity over said opening with said cavity communicating with said
opening and with said first seal and forming said vacuum in said cavity with a vacuum
source that communicates with said cavity through an opening in said guide member;
and
- (d) observing a level of said vacuum after a period of time.

13. (currently amended) The method of claim 10, further comprising:
~~inserting running~~ a cable through [[an]] said opening in said guide member
so that said cable extends through said fuselage of said aircraft;
forming a second seal between said guide member and said cable;
applying a vacuum to said second seal; and
observing a level of said vacuum after a period of time.

14. (original) The method of claim 13, wherein applying a vacuum to said second seal includes positioning a body of a testing tool having a cavity over said opening with said cavity communicating with said opening and with said second seal and forming a vacuum in said cavity with a vacuum source.

15. (original) The method of claim 10, wherein said cable has a diameter of about 0.5 inches or less.

16-18. (cancelled)